

Hypersonic aerothermoelastic response simulation using CFD/CSD approach

Zhi Chen¹, Jinan Lv¹, Bangcheng Ai¹, Jijun Yu¹

¹ China Academy of Aerospace Aerodynamics (CAAA), Beijing, China, 100074
madas1@126.com

Abstract: In this paper, a coupling numerical simulation technology which combined computational fluid dynamics (CFD) method with computational structure dynamic (CSD) is developed. The aerodynamic heating is calculated using CFD method. The thermal modal under aerodynamic heating is simulated using CSD method and the influence of aerodynamic heating is discussed. The aerothermoelastic response problem of an elastic hypersonic vehicle wing under Ma 5 flight condition is analyzed using CFD/CSD approach. The computed results indicate that using CFD/CSD simulation approach to predict aerothermoelastic character of hypersonic vehicle is feasible and credible.

Key Words: *CFD/CSD, aerothermoelastic response, hypersonic*

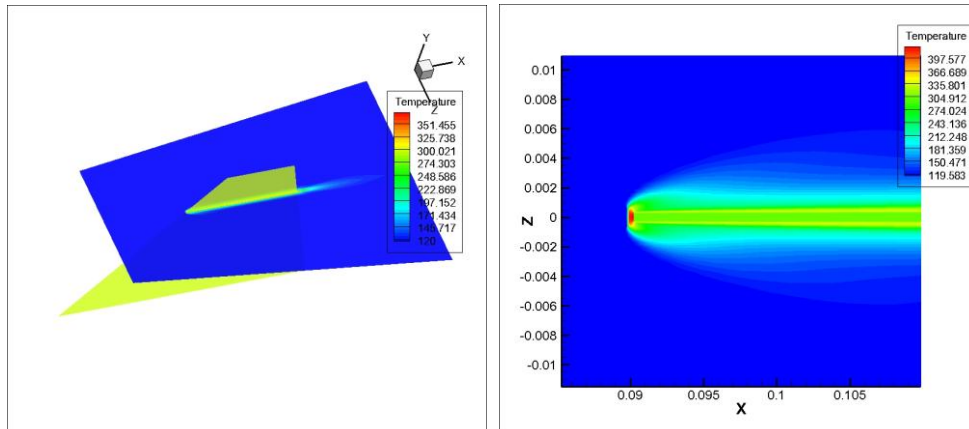


Figure 1 Film temperature contour

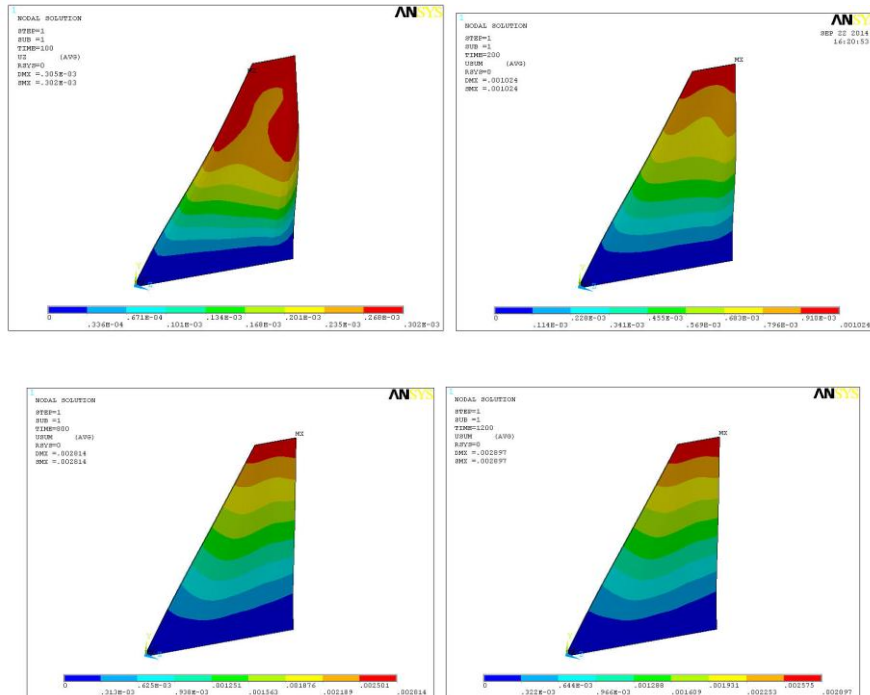


Figure 2 Structure thermal deformations with aerodynamic heating progressing

References

- [1] Peebles C, The X-43A flight research program: lessons learned on the road to Mach 10[M]. Washington, D. C. *AIAA Inc.*, 2007.
- [2] YANG Bing-yuan SHI Xiao-ming LIANG Qiang, Investigation and development of the multi-physics coupling dynamics on the hypersonic winged missiles[J]. *STRUCTURE & ENVIRONMENT ENGINEERING*, 2008, 35(5).
- [3] Yang Chao, Xu Yun, Xie Changchuan, Review of studies on aeroelasticity of hypersonic vehicles. *ACTA AERONAUTICA ET ASTRONAUTICA SINICA*, 2010, 31(1).
- [4] Thuruthinattam B J, Friedmann P P, Powell K G, et al, Aeroelasticity of a generic hypersonic vehicle[J]. *AIAA-2002-1209*.
- [5] Mcnamara J J, Thuruthinattam B J, Friedmann P P, et al, Hypersonic aerothermoelastic studies for reusable launch vehicles[J]. *AIAA-2004-1590*.
- [6] P.P. Friedmann, J.J. McNamara, B.J. Thuruthimattam and K.G. Powell, Hypersonic aerothermoelasticity with application to reusable launch vehicles, *AIAA 2003-7014*.
- [7] B.J. Thuruthimattam, P.P. Friedmann, J.J. McNamara and K.G. Powell, Modeling approaches to hypersonic aerothermoelasticity with application to reusable launch vehicles, *AIAA-2003-1967*.
- [8] Frederick W. Gibson, Flutter investigation of models having the planform of the North American X-15 airplane wing over a range of mach numbers from 0.56 to 7.3, *NASA TM X-460*.